AECMA STANDARD NORME AECMA AECMA NORM

prEN 4376

Edition P 1

December 1997

PUBLISHED BY THE EUROPEAN ASSOCIATION OF AEROSPACE INDUSTRIES (AECMA) Guiledelle 94 - B-1200 BRUXELLES - Tel. (+32) 2 775 8110 - Fax. (+32) 2 775 8111

ICS:

Descriptors:

ENGLISH VERSION

Aerospace series Heat resisting alloy NI-PH2601 (NiCr19Fe19Nb5Mo3) Solution treated and precipitation treated Bar and section $D_e \leq 200 \text{ mm}$

Série aérospatiale Alliage résistant à chaud NI-PH2601 (NiCr19Fe19Nb5Mo3) Mis en solution et précipité Barres et profilés $D_e \leq 200$ mm Luft- und Raumfahrt Hochwarmfeste Legierung NI-PH2601 (NiCr19Fe19Nb5Mo3) Lösungsgeglüht und ausgehärtet Stangen und Profile $D_e \leq 200$ mm

This "Aerospace Series" Prestandard has been drawn up under the responsibility of AECMA (The European Association of Aerospace Industries). It is published on green paper for the needs of AECMA-Members. It has been technically approved by the experts of the concerned Technical Committee following comment by the Member countries.

Subsequent to the publication of this Prestandard, the technical content shall not be changed to an extent that interchangeability is affected, physically or functionally, without re-identification of the standard.

After examination and signature of the AECMA Standard Checking Centre (NPS) and formal agreement of the Official Services of the Member countries it will be submitted as a draft European Standard to CEN (European Committee for Standardization) for formal vote.

Edition approved for publication	Comments should be sent within six months	C5 Chairman
	after the date of publication to	
1997-12-31	AECMA	Mr Evetts

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0 Introduction

This standard is part of the series of EN metallic material standards for aerospace applications. The general organization of this series is described in EN 4258.

This standard has been prepared in accordance with EN 4500-3.

1 Scope

This standard specifies the requirements relating to:

Heat resisting alloy NI-PH2601 (NiCr19Fe19Nb5Mo3) Solution treated and precipitation treated Bar and section $D_e \le 200 \text{ mm}$

for aerospace applications.

2 Normative references

This European Standard incorporates by dated or undated reference provisions from other publications. These normative references are cited at the appropriate places in the text and the publications are listed hereafter. For dated references, subsequent amendments to or revisions of any of these publications apply to this European Standard only when incorporated in it by amendment or revision. For undated references the latest edition of the publication referred to applies.

- EN 2043 Aerospace series Metallic materials General requirements for semi-finished product qualification (excluding forgings and castings) ¹⁾
- EN 3235-3 Aerospace series Heat resisting alloys Wrought products Technical specification -Part 3: Bar and section ¹⁾
- EN 4258 Aerospace series Metallic materials General organization of standardization Links between types of EN standards and their use ¹
- EN 4500-3 Aerospace series Metallic materials Rules for drafting and presentation of material standards Part 3 : Specific rules for heat resisting alloys ¹)

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4	Material designatio			<u>_</u>		- المحملة	resisting all		601 /880-	105-101	EMa21			
	Material designatio	1			1	Heat	T	-			5M03)		Т	
2		Eleme	nt	С	Si	Mn	P	S	Al	В	Co	Cr	Cu	
	Chemical	min.		0,02	-	_	-	_	0,20	20 *)	_	17,0	-	
	composition	max.		0,80	0,35	0,35	0,015	0,015	0,80	60 *)	1,00	21,0	0,30	
	%	Eleme	nt	Fe	Mg	Мо	Nb+Ta	Ti	Ag	Bi	Ca	Pb	Ni	
		min.		16,5	-	2,80	4,80	0,70	-		-	_	Base	
		max.		20,5	0,010	3,30	5,50	1,15	5 *)	1 *)	0,010	5 *)		
3	Method of melting			Consumable electrode remeited										
4.1	Form	Bar and section												
4.2	Method of producti	on			Wrought									
4.3	Limit dimension(s)		mm					D _e ≤						
5	Technical specifica	tion						EN 32	235-3					
6.1	Delivery condition			Solution treated Solution treated and precipitation						pitation tre	ated			
	Heat treatment			940 °C $\leq \theta \leq$ 980 °C/t = 1 h/AC or faster					940 °C $\leq \theta \leq$ 980 °C/t = 1 h/AC or faster + θ = 720 °C ± 10 °C/t = 8 h/FC at 50 °C per h $\leq \theta \leq$ 60° C per h ¹) to θ = 620 °C ± 10 °C/t = 8 h/AC					
6.2	Delivery condition	ode		W					U					
7	Use condition			Solut	Solution treated and precipitation treated				Delivery condition					
	Heat treatment	Delivery condition- $+\theta = 720$ °C ± 10 °C/t = 8 h/FC at-50 °C per h $\leq \theta \leq 60^{\circ}$ C per h 1)-to $\theta = 620$ °C ± 10 °C/t = 8 h/AC												
	Characteristics													
8.1 8.2	Test sample(s)			See EN 3235-3										
8.3	Test piece(s) Heat treatment			See EN 3235-3 Use condition										
9	Dimensions concer	ned	mm											
10	Thickness of cladd		%	D _e ≤ 200										
11	each face Direction of test pie	ce		L ²⁾					LT ²)					
12	Temperature	θ	°C	A	mbient		650		Ar	mbient		650		
13	Proof stress	R _{p0,2}	MPa		1 035		860		1	035		860		
14	T Strength	Rm	MPa		1 270		1 000		1	1 240		960		
15	Elongation	A	%		≥ 10		≥ 9			≥ 10		≥ 9		
16	Reduction of an	ea Z	%	≥ 15			≥ 15		≥ 8			≥ 8		
17	Hardness			≥ 331 HB										
18	Shear strength	Rc	MPa	_										
19	Bending	k	-	_										
20	Impact strength			-										
21	Temperature	θ	°C											
22	Time		h	t _R ≥ 23										
23	Stress	σa	MPa	-										
24	C Elongation	a	%	-										
25	Rupture stress Elongation at	 	MPa											
26	rupture	A	%	≥ 5										

*) 1) 2) 3)

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Notes (see line 98)